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SECTION 1 - GENERAL

1. Description and Characteristics

Wing span

Length
Height
Propeller ground clearance
(with front tyre and oleo-leg deflated):
positive

WING

The "Jodel" type wing is of the single spar type, covered with DACRON fabric.

Aspect ratio

Dihedral (wing-tip)

Chord (rectangular section)

Area

5.35
14°
67.4"
146,39 sq.ft

AILERONS

Total area 12.4 square feet
Deflection angles (see page 1.17)

The ailerons are controlled from the control column, by means of bell cranks, cables and pulleys, and are statically balanced.

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WING FLAPS

Total area

7.2 square feet

The flaps are manually operated by a lever located between the two front seats. They can be locked in three different positions.

- 1. Retracted 2. Take-off $(15^{\circ}_{-5^{\circ}}^{+0^{\circ}})$ (0.6 inches)
- 3. Landing $(60^{\circ+0^{\circ}})$ (0.6 inches)

NOTE: When set to the take-off or landing positions, a play of 0.6 inches (measured at the trailing edge) is normal.

TAILPLANE

Total area

31 square feet

The one-piece moving tailplane unit, statically balanced, is controlled by means of cables, and is equipped with a metal automatic anti-balance tab. The tab control is on the tunnel between the front seats and an index is graduated from 0 to 10.

- 0 = nose fully down
- 10 = nose fully up

Tailplane deflection angles (see page 1.17)
Anti-tab area
Anti-tab deflection angles (see page 1.17)

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FIN AND RUDDER

Rudder area

6.8 square feet

The rudder is conventionally controlled by means of a rudder bar and cables.

Rudder deflection angles

(see page 1.17)

LANDING GEAR

Fixed, tricycle type, with fairings and longstroke oleo legs. The three wheels are of identical type Removal of the wheel spats will considerably reduce level speeds and rates of climb.

The front landing gear is connected to the rudder bar by means of spring rods. During flight, the front wheel is automatically locked in alignment with the aircraft axis (oleo-leg extended).

Track
Wheel base
Whell size
Tyres

8 ft. 6 ins. 5 ft. 5 ins. 380 x 150

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Tyre Pressures (p.s.i.)	Front Rear	25 28	
Oleo leg strokes	Front Rear	6.28" 7.08"	
Oleo leg inflating pressures (p.s.i.)	Front Rear	64 78	

Fluid used - SHELL FLUID 4
BP AERO-HYDRAULIC NO. 1

BRAKES

Hydraulic braking system (independant on each wheel)

Braking is obtained by pushing the rudder bar pedals fully forward (from front seats)

The handbrake operates the brakes on both main wheels.

PARKING : Chocks must be used

HYDRAULIC FLUID : MIL.H.5606-A

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POWER PLANT

Engine: LYCOMING 0-360-A3A

4 horizontally opposed cylinders, direct drive, aircooled engine.

Max. contiuous RPM

Compression ratio

Max. cylinder head temp.

Max. cylinder temp

Direction of rotation

Firing order

2700

8.5:1

260°c.

160°c.

Clockwise

1:3:2:4

OIL SYSTEM

Wet	oil sump	(idling)	6 quarts
Oil	pressure		25 psi
011	grades:	(normal) above 15°c 30°C to -20°C	64 to 90 psi SAE 50 (100) SAE 40 (80)

Max oil temperature 118°c.

ELECTRICAL SYSTEM

A red warning lamp indicates failure to charge of the alternator. This circuit is protected by a 40 amp fuse.

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FUEL

AVIATION TYPE FUEL

91/96 (MIN) OR 100/130 OR 115/145

Fuel pressure :

max: 8 psi desired: 3 psi

min: 0.5 psi

MAIN (REAR) FUEL TANK CAPACITY 24.4 IG(29 USG)

The fuel control cock is mounted on the instrument panel tunnel.

ENGINE CONTROLS

The engine controls include an ON/OFF carburettor heater control (pull-type) and a mixture control (yellow knob).

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PROPELLORS				-	
	1	2	3	4	_
Makers	SENSENICH	SENSENICH	SENSENICH	SENSENICH	
Туре	76 EM	76 EM	76 EM	76 EM	,
	8\$50.64	8\$50.68	8\$50.58	8\$50.54	
Diameter *	76 "	76 "	76 "	76 "	
P i tch	64 #	68 "	58 "	54 "	
Min Speed (Full throttle	3,				1
MSL, fixed pitch)	2300 RPM,	2250 RPM	2500 RPM	2500 RPM	

NOTE: Avoid continuous use between 2150 and 2350 RPM

No reduction in diameter is permitted.

In Addition :

Propellor HO-27-HM-180/138

(Ø: 1,80 m, pitch: 1,38 m)
Minimum speed: 2400 RPM No RPM limitation

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CABIN

Fitted with sliding jettisonable canopy, opening from rear to front.

The two front seats are adjustable (6 positions).

Front and rear seats are all fitted with fast-release safety belts.

Cabin Dimensions

Length	64 "
Width	43.4"
Height	48.5"

Air conditioning

Two individually adjustable (flow and direction) fresh air vents are located on the instrument panel.

Cabin heating and canopy demisting controls are also fitted.

Heating is provided by means of a heat exchanger fitted around the RH exhaust manofold.

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DESCRIPTION OF VARIOUS EQUIPMENT

a. STANDARD EQUIPMENT

Dual throttle controls (actuating pick-up (quuiq Mixture control (yellow knob) Carburettor heater Battery switch Magneto selector key Starter button Cabin ventilation Cabin heater control Canopy demist control Fuel cock (4 positions) Stall warning (audio) "SAFE FLIGHT 164" Handbrake Tab control Fuel gauge 0il temperature gauge Ammeter RPM gauge (with hours-gone indicator) Magnetic compass Ball-type inclinometer Airspeed indicator Altimeter Vertical speed indicator Oil Cooler with thermostatic valve Oil pressure indicator Fuel pressure indicator

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Warning lamps for : flaps
fuel reserve level (front & rear)
oil pressure

alternator Hook release

Circuit breakers for:

warning lamps
indicators
electric pump
stall warning
starters
services
alternator

Tow hook and hook release haudle Cylinder Head Temerature gauge

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b. OPTIONAL EQUIPMENT

OAT gauge (canopy mounted) OAT gauge (tansmission type) Compass (above instrument panel) Electrical compass repeater Mixture monitor unit Inlet pressure gauge 3-pointer precision altimeter (in feet) JAEGER clock Chronometer Vacuum pressure gauge (for blind flying panel) Suction DI Suction-driven Artificial Horizon Electrical AH (with switch and fuse) Instrument panel lighting (2 red lamps with dimmer switch) Heated pitot head, warning lamp and switch Turn and Slip indicator (electric, suppressed

BRITTAIN turn co-ordinator
Rotating Anti-collision beacon
VHF radio
ADF
VOR
ILS

HF Radio

DME

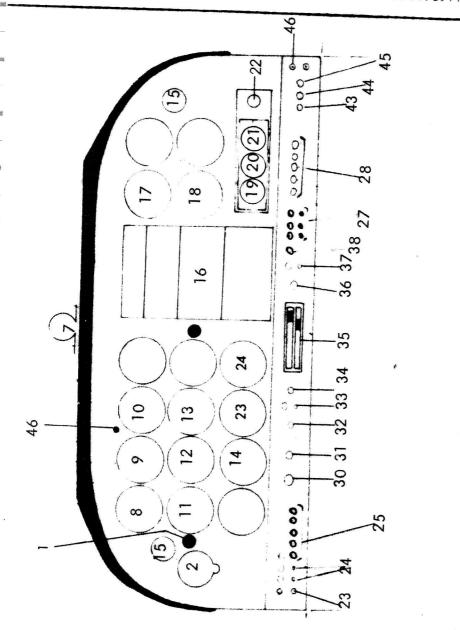
Marker beacon

Carburettor temperature gauge

RH and LH landing lamps, switch and fuse Navigation lamps

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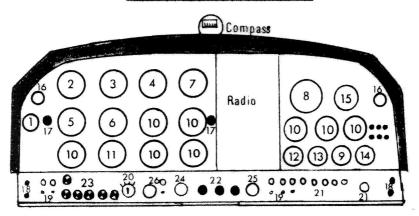


INSTRUMENT PANEL Nr 1

INSTRUMENT PANEL NR 1

- 1. Throttle control
- 2. Clock (option)
- 7. Magnetic compass (option) 8. Airspeed indicator
- 9. Artificial horizon (option) or compass
- 10. Altimeter
- 11. Turn and Slips indicator
- 12. Directionnal Indicator (option)
- 13. Vertical speed indicator
- 14. Suction gauge (option)
- 15. Fresh air vent
- 16. Radio (option) 17. Intake pressure (option)
- 18. RPM gauge
- 19. Rear tank gauge
- 20. Oil temperature 21. Anmeter
- 22. 40 A Fuse
- 23. Headset jacks (option) 24. Circuit breakers
- 25. Warning lights
- 26. Instrument panel lights (option) 27. Circuit breakers
- 28. Fuses
- 30. Magneto switches
- 31. Starter
- 32. Battery switch 33. Alternator switch
- 34. Mixture control
- 35. Heating controls
- 36. Carburetor heater
- Electric pump switch
- 43. 44. 45. Fuses
- 45. Hook release warning light

INSTRUMENT PANEL - Nr 2



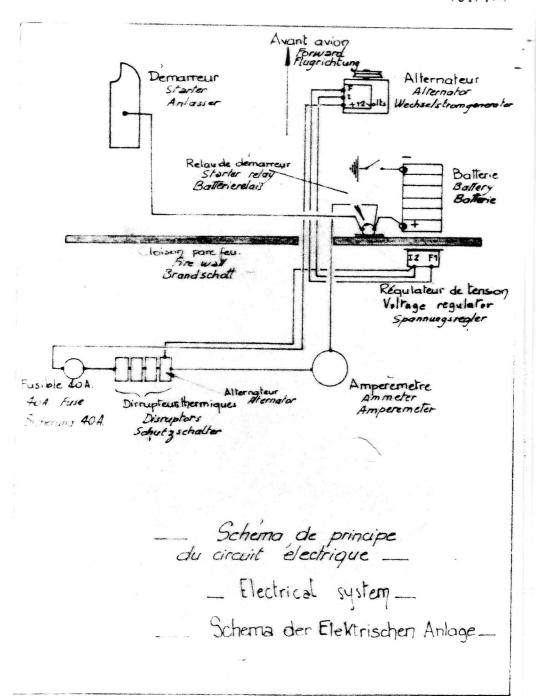
- 1 Stop watch (option)
- 2 Airspeed indicator
- 3 Artificial horizon
- 4 Altimeter
- 5 Turn/bank indicator
- 6 Directionnal (opt.)
- 7 Rate of climb
- 8 RPM indicator
- 9 Fuel pressure (opt.)
- 10 Options

- 23 Warning lights
- 24 Mixture control
- 25 Carburator heater
- 26 Battery and alternator switch

- 11 Vacuum gauge (opt.)
- 12 Oil temperature
- 13 Oil pressure
- 14 Amp. or voltmeter
- 15 Fuel gauge
- 16 Freshair vent
- 17 Throttle control
- 18 Radio jacks plugs
- 19 Switches-breakers
- 20 Magneto switch
- 11 Breakers
- 22 Heating/demisting

Flight Manual Edition Issue Flughandbuch Ausgabe nº 1 DR 400/180 R du: 15.7.1974 Avant aving (forward Flugrichtung) Circuit d'essence Fuel System .Carburateur Kraftstoffsystem (Moteur) Vergaser Pompe mécanique (Moteur). 1 Engine de Kraftstoffgrunge cloison pare feu Pompe électrique de secours (montée en serie) Elect emergency pumps Elekt. Benzinfunge Filtre-deconteur filter & drain et purge filter u Entwissenund Robinet Selecteur Enel Selector work Brandhahn Reservoir AR Remplissage 110 litres Rear tank 242 UK. gol du reservoir Either cap Hinterer Tank Holster Enfillstatzen - Purge / Drain jaujeur (fuel gange Kraftstoffangeige) Entwisera Mise à l'air libre sur dos de fuselage. Evel tank vent on top of fuelag Tankentligtung uber dem Rumpf

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